

AS3020: Aerospace Structures Module 6: Introduction to Variational Mechanics

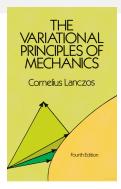
Instructor: Nidish Narayanaa Balaji

Dept. of Aerospace Engg., IIT-Madras, Chennai

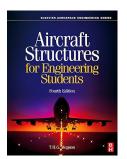
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Very good book to read Lanczos [1]



Chapters 4, 5 in Megson [2]

1. The Principle of Virtual Work

• The idea behind the equilibrium principle is that the sum of all forces acting on a body is zero:

$$\int_{\Omega} forces = 0.$$

• In module 3, we used this in conjunction with the Cauchy Stress Principle to obtain general governing equations for an elastic solid:

$$\sigma_{ij,j} + f_i = 0$$
, on Ω , + $b.c.s$ on $\partial \Omega$.

- The work done by any force f_i on a system as it goes from $u_i^{(0)}$ to $u_i^{(0)} + \delta u_i$ is written as $f_i \delta u_i$.
- The displacement field δu_i denotes a "virtual displacement", which is a mathematical idealization such that
 - it is small enough so as not to introduce changes in the force field;
 - it is compatible with any constraints that exist (B.C.s, for instance).

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 - it is small enough so as not to introduce changes in the force field;
 - it is compatible with any constraints that exist (B.C.s, for instance).

Under the stated assumptions, $\delta(\cdot)$ can be treated as a differential operator and we call it the **variational operator**.

1. The Principle of Virtual Work

• The work done by a virtual displacement field is termed as the *virtual work*. For a system under a certain force-field, the *virtual work* is a property of the system since so further deformation needs to be done..

Principle of Virtual Work

The virtual work of a system at equilibrium is zero.

- The *principle of virtual work* is merely a restatement of the principle of equilibrium, but it sometimes provides a more convenient analytical framework.
- Variational Mechanics is sometimes also referred to as analytical mechanics.

1.1. The Elastic Solid

The Principle of Virtual Work

• For the elastic solid, the principle of virtual work may be mathematically expressed as

$$\int_{\Omega} \underbrace{\sigma_{ij,j} \delta u_i}_{(\sigma_{ij} \delta u_i), j - \sigma_{ij} \delta u_{i,j}} + \int_{\Omega} f_i \delta u_i = 0.$$

• Applying Gauss divergence to the first term in the above we get,

$$\int_{\partial\Omega} \underbrace{\sigma_{ij} n_j}_{\text{surface traction } t_i} \delta u_i - \int_{\Omega} \sigma_{ij} \delta u_{i,j} + \int_{\Omega} f_i \delta u_i = 0.$$

• Due to stress tensor symmetry, the following equality holds:

$$\sigma_{ij}\delta u_{i,j} = \sigma_{ij}\delta\underbrace{\left(\frac{u_{i,j} + u_{j,i}}{2}\right)}_{E_{i,i}} = \sigma_{ij}\delta E_{ij}.$$

• So the principle of virtual work reads,

$$\int_{\Omega} \sigma_{ij} \delta E_{ij} = \int_{\Omega} f_i \delta u_i + \int_{\partial \Omega} t_i \delta u_i$$



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• So the principle of virtual work reads,

$$\begin{array}{c}
\text{"internal"} \\
\text{contributions}
\end{array}
\longrightarrow \int_{\Omega} \sigma_{ij} \delta E_{ij} = \int_{\Omega} f_i \delta u_i + \int_{\partial \Omega} t_i \delta u_i \leftarrow \begin{bmatrix}
\text{"external"} \\
\text{contributions}
\end{bmatrix}$$

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1.1. The Elastic Solid

The Principle of Virtual Work

ullet The formula above is valid in the general case but further simplifications are possible for the non-dissipative solid. Here we know that a strain energy density $\mathcal U$ exists such that

$$\sigma_{ij} = \frac{\partial \mathcal{U}}{\partial E_{ij}}, \text{ and } \int_{\Omega} \mathcal{U} = U.$$

• Substituting this for the internal components we get,

$$\int_{\Omega} \sigma_{ij} \delta E_{ij} = \int_{\Omega} \frac{\partial \mathcal{U}}{\partial E_{ij}} \delta E_{ij} = \int_{\Omega} \delta \mathcal{U} = \delta U.$$

 \bullet Denoting the external contributions by Π such that

$$\delta\Pi = \int_{\Omega} f_i \delta u_i + \int_{\partial\Omega} t_i \delta u_i,$$

the principle of virtual work can be simply written as,

$$\delta(\mathbf{U} - \Pi) = 0.$$

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• Note that while the strain energy U is fully described by the system,

II is loading-state dependent.

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1.2. Interpretation

The Principle of Virtual Work

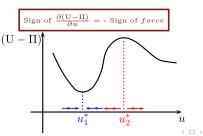
- $\delta(U \Pi) = 0$ can be stated as For a system in equilibrium, the variation of total work done is zero.
- Intuitively, this means that the equilibrium deformation field of a system extremizes the quantity $U-\Pi$:

 Out of all possible deformation fields that satisfy the constraints, the system deforms in a way that extremizes the quantity $U-\Pi$.

Stability

• Suppose u^* is the equilibrium field, the "surplus energy" $U - \Pi$ in its neighborhood governs its stability.

- Consider the 1D example here with two extremal (equilibrium) points: u_1^* and u_2^* .
- The local behavior of the function $U \Pi$ governs the stability of the equilibria.



2. Beam Bending: A Motivating Problem Setting

• The kinematic and stress descriptions of a symmetric slender beam on the $(\underline{e}_1, \underline{e}_2)$ plane is:

$$u_1 = -X_2v', \quad u_2 = v, \quad E_{11} = -X_2E_yv'', \qquad \sigma_{11} = E_yE_{11}.$$

• The strain energy density in this case is

$$\mathcal{U} = \frac{E_y}{2} X_2^2 \left(v'' \right)^2.$$

ullet Integrating this over the section $\mathcal S$ we get the linear density

$$d\mathbf{U} = \frac{E_y I_{33}}{2} \left(v''\right)^2.$$

• Considering transverse body force (per unit length) f and some point force F_P , the external energy is given as,

$$\Pi = \int_0^\ell fv + F_P v(X_P).$$

• Combining the two and integrating over the length we have,

$$U - \Pi = F_P v(X_P) + \int_0^\ell \frac{E_y I_{33}}{2} (v'')^2 - fv.$$

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2.1. Calculus of Variations

Beam Bending: A Motivating Problem Setting

• Ignoring the $F_P v(X_P)$ term for the moment, the principle of virtual work is given as,

$$\delta(U - \Pi) = \delta \left(\int_{0}^{\ell} \frac{E_y I_{33}}{2} (v'')^2 - fv \right) = 0.$$

• This is a variational equation of the form,

$$\delta\left(\int_{\mathcal{D}} \mathcal{L}(v, v', v'', \dots)\right) = 0.$$

• Since δv are small quantities, we can apply Taylor's expansion on:

$$\int_{\mathcal{D}} \frac{\partial \mathcal{L}}{\partial v} \delta v + \frac{\partial \mathcal{L}}{\partial v'} \delta v' + \frac{\partial \mathcal{L}}{\partial v''} \delta v'' + \dots = 0.$$

• Applying integration by parts and observing that the variations $\delta v^{(n)}$ vanish at the boundaries, this simplifies as,

$$\int_{\mathcal{D}} \left(\frac{\partial \mathcal{L}}{\partial v} - \frac{d}{dx} \frac{\partial \mathcal{L}}{\partial v'} + \frac{d^2}{dx^2} \frac{\partial \mathcal{L}}{\partial v''} + \dots \right) \delta v = 0.$$

 $f_{\mathcal{D}} \setminus OV = dx OV' = dx^2 OV''$ Balaji, N. N. (AE, IITM)

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2.1. Calculus of Variations

Beam Bending: A Motivating Problem Setting

• Since the integral condition needs to be satisfied for all kinds of variations $\underline{\delta u}$, the term within the parens must be zero:

$$\frac{\partial \mathcal{L}}{\partial v} - \frac{d}{dx} \frac{\partial \mathcal{L}}{\partial v'} + \frac{d^2}{dx^2} \frac{\partial \mathcal{L}}{\partial v''} + \dots = 0.$$

• This is known as *Euler-Lagrange Equations* and forms the basis of variational mechanics. It is the *functional analog* of function extremization.

Regular Calculus

- Find $x \in \mathbb{R}$ such that $f(x) : \mathbb{R} \to \mathbb{R}$ is **extremized**.
- First order optimiality condition: $\frac{df}{dx} = 0$.
- Second order optimality: $\frac{d^2f}{dx^2} :> 0(\min), < 0(\max).$

Calculus of Variations

- Find $v(x) : \mathbb{R} \to \mathbb{R}$ to extremize functional $J = \int \mathcal{L} dx$.
- First order optimality condition: $\frac{\partial \mathcal{L}}{\partial v} \frac{d}{dx} \frac{\partial \mathcal{L}}{\partial v'} + \frac{d^2}{dx^2} \frac{\partial \mathcal{L}}{\partial v''} + \dots = 0.$
- Second order optimality: not so trivial.

2.2. Variational Derivation of Euler Bernoulli Beam Theory

Beam Bending: A Motivating Problem Setting

• Returning to the slender beam, we have as the Lagrangian,

$$\mathcal{L} = \frac{E_y I_{33}}{2} (v'')^2 - fv.$$

• Applying the Euler-Lagrange Equations we obtain,

$$\underbrace{-f}_{\frac{\partial \mathcal{L}}{\partial v}} + \frac{d^2}{dX_1^2} \underbrace{\left(E_y I_{33} v''\right)}_{\frac{\partial \mathcal{L}}{\partial v''}} = 0.$$

• For a beam with uniform properties (E_yI_{33} constant along the beam), the governing equations may be written as

$$E_y I_{33} v^{\prime\prime\prime\prime} - f = 0,$$

which is precisely what we expect from Euler-Bernoulli Beam Theory.

So far we've just reinvented the wheel and not really shown an example where the Variational approach really shines. We will do this next.

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3. Tutorial Example

- Consider the following fixed-free beam along with a spring support at the end.

 Remember the
- The energy quantities are,

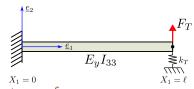
$$U = \int_{0}^{\ell} \frac{E_{y} I_{33}}{2} (v'')^{2} dX_{1} + \frac{1}{2} k_{T} v_{T}^{2}, \qquad \Pi = F_{T} v_{T}.$$

integral here.

• Since the contributions from the load as well as the spring are from the boundaries, the bulk Equations of Motion (EoM) remains unchanged as: $E_y I_{33} v'''' = 0$.

Boundary Conditions

Clamped End v = 0, v' = 0, $X_1 = 0$ Loaded End $v = v_T$, v'' = 0, $X_1 = \ell$



• Solution in $X_1 \in (0, \ell)$ can be written in terms of v_T as:

$$v(X_1) = \frac{v_T}{2\ell^3} X_1^2 (3\ell - X_1).$$

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3. Tutorial Example

• Substituting this back into the energy quantities and integrating it yields,

$$U - \Pi = \underbrace{\frac{3E_y I_{33}}{2\ell^3} v_T^2}_{\int_0^\ell \frac{E_y I_{33}}{2} (v'')^2} + \frac{k_T}{2} v_T^2 - F_T v_T.$$

• Extremization of this quantity is trivial since everything just depends on a single unknown scalar, v_T . So we have,

$$\delta(\mathbf{U} - \mathbf{\Pi}) = \delta v_T \frac{\partial}{\partial v_T} (\mathbf{U} - \mathbf{\Pi}) = \delta v_T \left(\left(\frac{3E_y I_{33}}{\ell^3} + k_T \right) v_T - F_T \right).$$

• Setting this to zero for all δv_T implies

$$v_T^* = \frac{F_T}{k_T + \frac{3E_y I_{33}}{\ell^3}},$$

which is the equilibrium deflection.

• Plugging this back into the solution $v(X_1)$ above yields the full deformation shape.

4. Bending of Axially Loaded Beams

• We now draw upon the kinematic description from before for an axially loaded beam. The kinematics are,

$$u_1 = -X_2 v'(X_1), \quad u_2 = v(X_1).$$

• The corresponding axial strain is,

$$E_{11} = u_{1,1} + \frac{1}{2} \left(u_{1,1}^2 + u_{2,1}^2 \right)$$

= $-X_2 v'' + \frac{1}{2} \left(X_2^2 (v'')^2 + (v')^2 \right).$

• Considering a loading case that leads to a uniform axial stress (σ_{11}) of $\frac{P}{A}$, the strain energy can be written as

$$U = \int_{0}^{\ell} \left(\int_{\mathcal{S}} \frac{P}{A} E_{11} + \frac{E_y}{2} E_{11}^2 dA \right) dX_1$$
Retaining up to $\mathcal{O}(v^2)$ terms
$$= \int_{0}^{\ell} \frac{E_y I_{33}}{2} \left(1 + \frac{P}{E_y A} \right) (v'')^2 + \frac{P}{2} (v')^2 dX_1$$

4. Bending of Axially Loaded Beams

• Considering transverse loads $f(X_1)$, the external work done is

$$\Pi = \int_{0}^{\ell} fv dX_1.$$

• The Lagrangian density is

$$\mathcal{L} = U - \Pi = \frac{E_y I_{33}}{2} \left(1 + \frac{P}{E_y A} \right) (v'')^2 + \frac{P}{2} (v')^2 - fv.$$

• The Euler-Lagrange equations (for stationarity) are

$$E_y I_{33} \left(1 + \frac{P}{E_y A} \right) v'''' - Pv'' - f = 0$$

• We expect the term $\frac{P}{E_{v}A}$ to be small in general.



4.1. Cables and Beams

Bending of Axially Loaded Beams

• The first simple case we will consider is the case of extremely small cross-sections. Here it can be said that

$$A \sim \mathcal{O}(\epsilon^2), \qquad I_{33} \sim \mathcal{O}(\epsilon^4).$$

• For sufficiently small ϵ , contributions from I_{33} can be negligible. In general, when

$$E_y I_{33} \ll P$$
,

the following approximation may be made, wherein the equations governing transverse deflection become:

$$Pv'' + f = 0.$$

• This governs the deflection of cables wherein the axial tension is the primary source of stiffness. This is also a second order differential equation, unlike beam bending, which is a fourth order equation.

4.2. Column Buckling

Bending of Axially Loaded Beams

• Another simplification is possible for the absence of transverse loads. Here the general equations read:

$$E_y I_{33} v'''' - \frac{P}{1 + \frac{P}{E_y A}} v'' = 0.$$

• Since $\frac{P}{E_{vA}} << 1$, we have

$$\frac{P}{1 + \frac{P}{E_y A}} = P - \frac{P^2}{E_y A} + \frac{P^3}{E_y A^2} - \dots = P,$$

which leads to

$$E_y I_{33} v'''' - P v'' = 0.$$

ullet For compressive loading $(P \to -P)$, this is the classical buckling equation

$$E_y I_{33} v'''' + P v'' = 0.$$

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• Finding values of P where the trivial solution v = 0 is unstable leads to a Sturm-Liouville problem which will be treated in the **next module**.

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References I

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- [2] T. H. G. Megson. Aircraft Structures for Engineering Students, Elsevier, 2013. ISBN: 978-0-08-096905-3 (cit. on p. 2).